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NAVAL OCEAN RESEARCH AND DEVELOPMENT ACTIVITY NSTL S--ETC F/G 9/2  
REDUCING LASER PROFILES ON A TABLETOP COMPUTER. (U)

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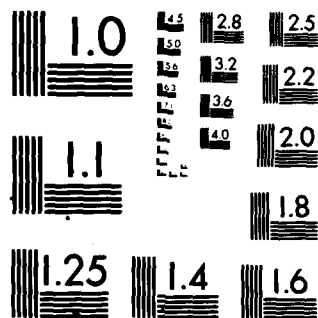
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MICROCOPY RESOLUTION TEST CHART  
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**LEVEL II**

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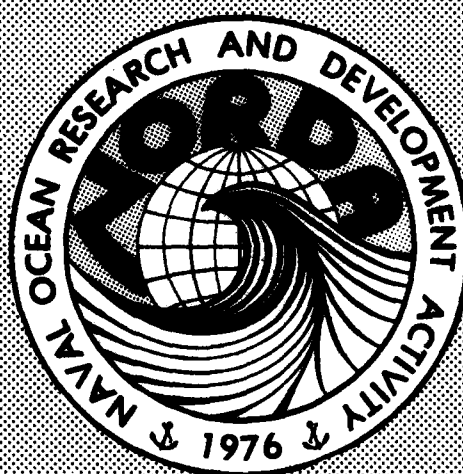
NORDA Technical Note 77

See 14731

Naval Ocean Research  
and Development Activity  
NSTL Station, MS 39529

## Reducing Laser Profiles on a Tabletop Computer

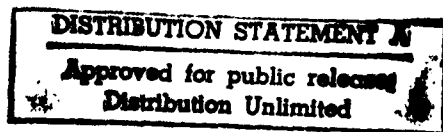
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# ABSTRACT

Standard numerical techniques for removing aircraft motion and discontinuities from airborne laser profiles are adapted to a desk-top computer. Because such a computer is much slower than a larger machine, analog active filters replace the numerical (Hamming) filters normally used. The computer programs are in BASIC, and listings are provided in an Appendix. Several examples of data editing procedures are given.

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## BACKGROUND

Airborne laser profiling remains by far the most rapid method of obtaining some measure of the "roughness" of a field of ice. This paper will deliberately not deal with the applications of laser profiles, but will describe a specific method of obtaining "clean" profiles from raw data. The reader is referred to Ketchum (1971) for a general description of the profiling system, a Spectra-Physics Geodolite 3A.

Ketchum (1971) introduced airborne laser profiles to the Arctic remote sensing community, and Hibler (1972) devised a computer-based scheme for removing unwanted "aircraft motion" from the raw data. This scheme has been used and modified in subsequent applications (Holyer et al., 1977; Lohanick, 1979), but not greatly changed.

This paper will describe a specific data reduction system written at NORDA, centered around Hewlett-Packard hardware and firmware. It is based on the Hibler three-step filter process, but is accomplished on a mini-computer dedicated to this task, and with some steps which greatly reduce overall data reduction time.

## EVALUATION OF RAW PROFILE DATA

Profile data are recorded in the field on FM analog tapes. A time code channel is provided for comparison of the data with annotations on various aircraft logs, and for correlation with other airborne sensors which also record time. These analog tapes must first be converted to digital form (digitized) for handling in the computer. The initially digitized data from the analog field tapes will hereafter be referred to as the "raw" data.

For those unfamiliar with the "look" of raw laser profile data, Figure 1 is provided and annotated to point out relevant features. The vertical exaggeration in this plot is about 300:1.

## SAMPLING RATE

The sampling rate (number of points digitized per second) for this type of data has historically given an actual spacing of about 1 m (see Hibler, 1972; Holyer et al., 1977) on the terrain. Hibler's (1972) scheme leaves intact all data that have a wavelength less than about 127 m, although the power spectrum of the topside of sea ice has never specifically been discussed in the short wavelength region (see Hibler and LeSchack, 1972).

## AIRCRAFT MOTION

Long-wavelength undulations in the data are due to the motion of the aircraft carrying the laser. As the aircraft slowly rises and falls in altitude, and makes turns, the apparent altitude measured by the profiler varies. This result of aircraft instabilities must be removed if the actual terrain profile is to be recovered.

## DISCONTINUITIES

Discontinuities in the apparent profile, such as at points A and B in Figure 1, can be caused either deliberately or accidentally by the profiling system. Causes for these jumps in the apparent profile are discussed in Holyer et al. (1977). Figure 2 shows the results of not removing these discontinuities. Nonexistent terrain features appear in final profile, because the computer recognizes only long-wavelength variations as aircraft motion.

Hibler's 1972 scheme for aircraft motion removal is adequate, and lends itself to numerical or analog processes, as will be discussed below. Removal of discontinuities is essential to the data reduction. But, as pointed out by Holyer et al. (1977), no satisfactory method has yet been devised for reliably eliminating these discontinuities automatically (i.e., without the intervention of a trained analyst). Therefore, some sort of interactive system must be devised to allow a trained operator to choose sections of data visually, as well as to remove undesirable discontinuities. The alternative to interactive graphics is a laborious process involving hand annotation of computer-generated plots and print listings (Lohanick, 1979).

Figure 3 is included to show a data record for which reliable automatic algorithms for locating discontinuities would be difficult to construct. The data were recorded with four times the vertical sensitivity of those in Figure 1. The circled area is referred to in a later section.

#### RELEVANT CAPABILITIES OF THE COMPUTER HARDWARE

All elements of the hardware system used in the present application are sketched in Figure 4. The computer is a Hewlett-Packard (HP) 9845B with 186 kilobytes of internal memory, a CRT for printing, data display and graphics, an internal thermal printer and two data cartridge (tape) drives. The 9845B has four interface busses for peripherals.

CRT graphics is done on a 560 x 455 dot matrix. Any plot generated on the CRT can be printed on the internal printer with one command. It is also possible to store any plot on a mass-storage device such as a flexible disk or tape cartridge.

The HP2240A Measurement and Control Processor contains both an A/D converter and a D/A converter, whose uses will be described later. Both converters have a resolution of 5 mV in a range of +10,000 mV (approximate range). The 2240A is directly compatible with the 9845B. The data transfer rate for A/D (12-bit) conversion is about 100 Hz. The HP9885 Flexible Disk drives use seven-inch disks (floppies) with a storage capacity of about 500 K bytes.

Two double Rockland active filters (Models 432 and 452-01) are used between the 2240A D/A output and A/D input to accomplish aircraft motion removal. They replace previously used numerical filters, which are far too slow in this application. A 30000-data point record, which would require 60 hours of time on this computer with the numerical filters, requires about 1.5 hours with active filters.

#### THE DATA REDUCTION PROCESS

##### DIGITIZING (PROGRAM "A/D")

Because of hardware and firmware constraints, a maximum of 90,000 regularly-spaced integer data points (16 bits) can be taken from an analog field tape at one time. Since the maximum sampling rate of the system is 100 12-bit samples/second, this amounts to 15 minutes of data, or about 50 nautical miles (at 200 knots aircraft speed).

A data tape is chosen and placed on the Ampex FR-1300 Tape Recorder. A suitable section of the tape is located by reading the time-code channel and comparing with aircraft logs.

The computer program used is named "A/D" and its action is diagrammed in Figure 5 (all programs are in BASIC, and complete listings are provided in Appendix A, with internal documentation). The program enables the HP2240A to read the analog tape, digitize the analog voltages, and make them available for reading by the HP9845B computer. For maximum data transfer rate (100 Hz) the 2240A constantly fills its output buffers with data, while the 9845B asynchronously empties them into internal memory. In this way, time-consuming synchronization is avoided.

Once the 90000 (or fewer) data points are read, digitizing ceases, and the data is immediately stored on a floppy. Because firmware restricts any file to 32767 elements ( $2^{15}-1$ ), a file length of 30000 is chosen. Thus, a full run will fill three files, each containing five minutes of data.

During program execution, the operator presses one key to begin or end the digitizing, and names the files which will store the raw profile data.

#### REFORMATTING AND PLOTTING (PROGRAM "PLOT")

Some reformatting of data is done to take advantage of special dense data file types which allow storage of eight full files (40 minutes of data) on each floppy. The data in one file are then plotted in small scale on the CRT and stored on disk. A hard copy of the CRT plot is obtained on the thermal printer (see Fig. 6), and is used for some annotation and time-scaling. The stored plot is used for editing the data as described below.

The small scale plot is the visual counterpart of a raw data file, and both are saved permanently.

Program "Plot" requires the operator to name the file which will store the plot.

#### EDITING (PROGRAM "EDIT")

Program "Edit", takes about 85% of total operator time, since it requires him to make judgments as to the quality and character of the data which has been digitized and plotted. He must also choose one of several editing procedures to remove undesirable discontinuities. At this point, the operator must be a trained analyst who has seen and been guided through, much raw profile data and its peculiarities. The operator's expertise will, of course, have a direct impact on the quality of the final product. Some examples given below will help to point this up.

To begin, the operator names the raw data file and its plot (file). The computer reads these into memory and graphics memory, respectively, and presents the small-scale plot (Fig. 6, again) on the CRT. The operator uses a small cursor (operated from the keyboard) to move to features he wishes to change or to observe more closely. He presses one key, and the CRT provides a large-scale plot of 60 data points in the immediate vicinity of the current point of interest (Fig. 7). This large-scale plot will be called an edit window. The vertical exaggeration in the edit window is about 2:1.

Five different editing procedures are available. Two of these require the operator to mark the beginning and end of the data record (BOR, EOR), which are not defaulted. The three editing procedures which affect the data are:

1. Straight-line tie (see Fig. 8a). This procedure allows the operator to remove a sharp discontinuity and replace it with a horizontal straight line. It is

appropriate when the discontinuity is a "180° phase shift" (an end-of-scale jump back to center scale, provided by the laser hardware to avoid having to record voltages outside a selected range, usually  $\pm 1.5$  volts), or a spike which may be caused by spurious effects during recording or digitizing.

The operator places the cursor to the immediate left of the discontinuity, presses a key to enter its position, moves the cursor to the present position of the end of the feature, presses the key, moves the cursor to the desired position of this second point (only up and down motion is permitted), and enters it. The CRT shows the result for approval or disapproval (Fig. 8b).

2. Straight-line slope (see Fig. 9a). The operator marks all the points as above. The computer leaves the character of the data in the interval between points 1 and 2 intact, but changes the slope (Fig. 9b).

3. Point-by-point fix (see Fig. 10). Allows the operator to move each point a desired distance up or down. This procedure is appropriate when there are several discontinuities in the edit window, and the data are considered reliable by the operator. Several steps in this process are shown in Figures 10b through 10d.

The product of this program is a tape file of all edit windows for a particular record of raw data (which may be a complete raw data file or portion of it, delineated by the BOR and EOR mentioned above). This program is the only step requiring great skill on the part of the operator.

#### REMOVING THE DISCONTINUITIES (PROGRAM "DISCON")

The operator provides the raw data file disk and the correct edit window file, and the computer automatically removes all discontinuities as prescribed in the editing procedure above. The operator provides a name for the updated file, which is stored again on a floppy. The original file is not destroyed. Edit window files are generally not saved, since they are intermediate products and are not needed beyond this stage.

#### AIRCRAFT MOTION REMOVAL (PROGRAM "8-IN-1")

The operator provides the program with the names of the source file(s) (the one(s) created in Program Discon above) and the result file(s). The program is written to handle up to eight pairs of files at a time.

Aircraft motion removal has previously been done with numerical Hamming filters (Hibler, 1972; Holyer et al., 1977; Ketchum, 1971; and Lohanick, 1979). The approximate number of multiplications to be done by one pass of a Hamming filter is the number of points in the record times the number of filter weights. For the low pass filter providing the aircraft motion envelope (see below), the number of filter weights for a sampling rate of 100 points/second is about 250, leading to over six million calculations for a 30000-point record. On the 9845B this would take many hours. In fact, for all three filters, estimated total run time for a five-minute record is about 60 hours, so we must use external active filters.

Data flow through a particular filter is diagrammed in Figure 11. Data are first read into an array in computer memory. Then one point at a time is read into the D/A converter and fed to the appropriate active filter. The output of the filter (connected to the A/D converter) is "simultaneously" read back into computer memory. Data are filtered in pseudo-real-time, since one complete cycle of D/A and A/D conversion is about six times as long as the actual data sampling interval.



Thus, a 30000-point record (5 minutes of data) takes about 30 minutes to filter. Filter cutoff frequencies must be set to about one-sixth of the values calculated for the numerical filters. The entire aircraft motion removal scheme is explained in Hibler (1972).

Since eight files (40 minutes of data) can be handled by the program at one time, a complete run takes about 12 hours, and is run unattended overnight. Operator time is a total of about 15 minutes in setting up. Result files, which are the final terrain profiles, are stored on disk.

One product of Program "8-in-1" is shown in Figure 12. This scaled plot (horizontal dotted lines are at 2 m intervals above the zero line) has a great amount of vertical exaggeration (about 1000:1). The operator uses this plot to look for any unusual features, such as ridges over 10 m high or large spikes below the zero line. Causes for these features must be found. If the cause is faulty editing, then editing must be repeated, but only for this particular error.

The profile is now in its final form on disc, and is ready to be used in any application of sea-ice terrain features such as roughness, ridge counts, ridge height distribution, or power spectral density.

A summary of all steps taken in the data reduction is shown in Figure 13, for a segment about 2 km long.

#### REFERENCES

Hibler, W. D. III (1972). Removal of Aircraft Altitude Variations from Laser Profiles of the Arctic Ice Pack, J. Geophys. Res., 77(30) p. 7190-7195.

Hibler, W. D. III and L. A. LeSchack (1972). Power Spectrum Analysis of Undersea and Surface Sea Ice Profiles. J. Glaciol. 11, 63, p. 345-356.

Holyer, I. J. J., P. Wadhams and R. T. Lowry (1977). An Interactive Graphics System for the Reduction of Airborne Laser Profiles of Sea Ice. Scott Polar Research Institute Tech. Rept. 77-1, Cambridge, England.

Ketchum, R. D., Jr. (1971). Airborne Laser Profiles of the Arctic Pack Ice, Remote Sensing of Environment 2, p. 41-52.

Lohanick, A. W. (1979). Airborne Laser Sea Ice Profiles Near a Drifting Camp, April 1977, NORDA Technical Note 49.

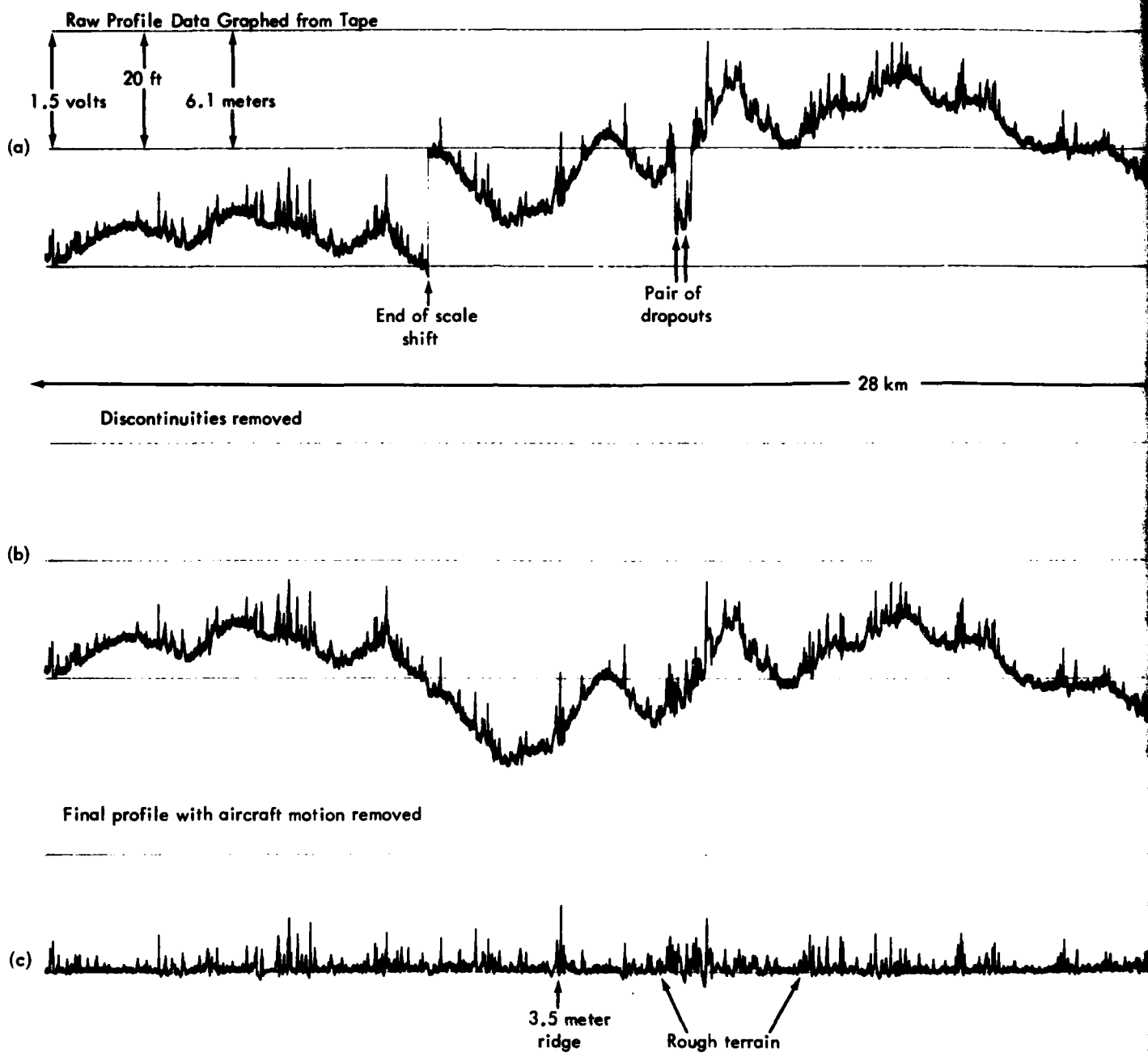


Figure 1. Laser sea-ice profile showing stages of discontinuity motion removal. Vertical scale exaggeration about

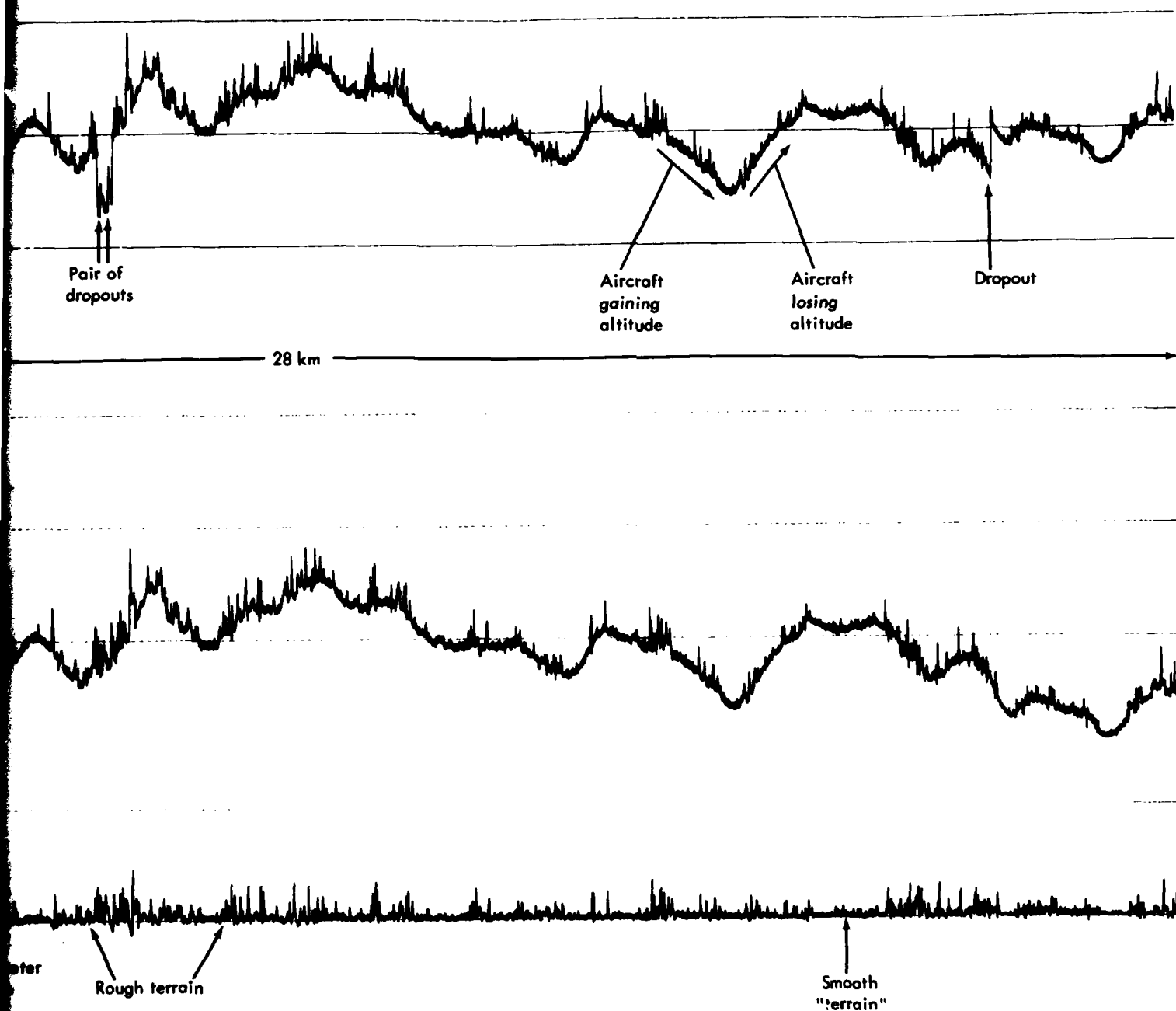


Figure 1. Laser sea-ice profile showing stages of discontinuity and aircraft motion removal. Vertical scale exaggeration about 300:1.

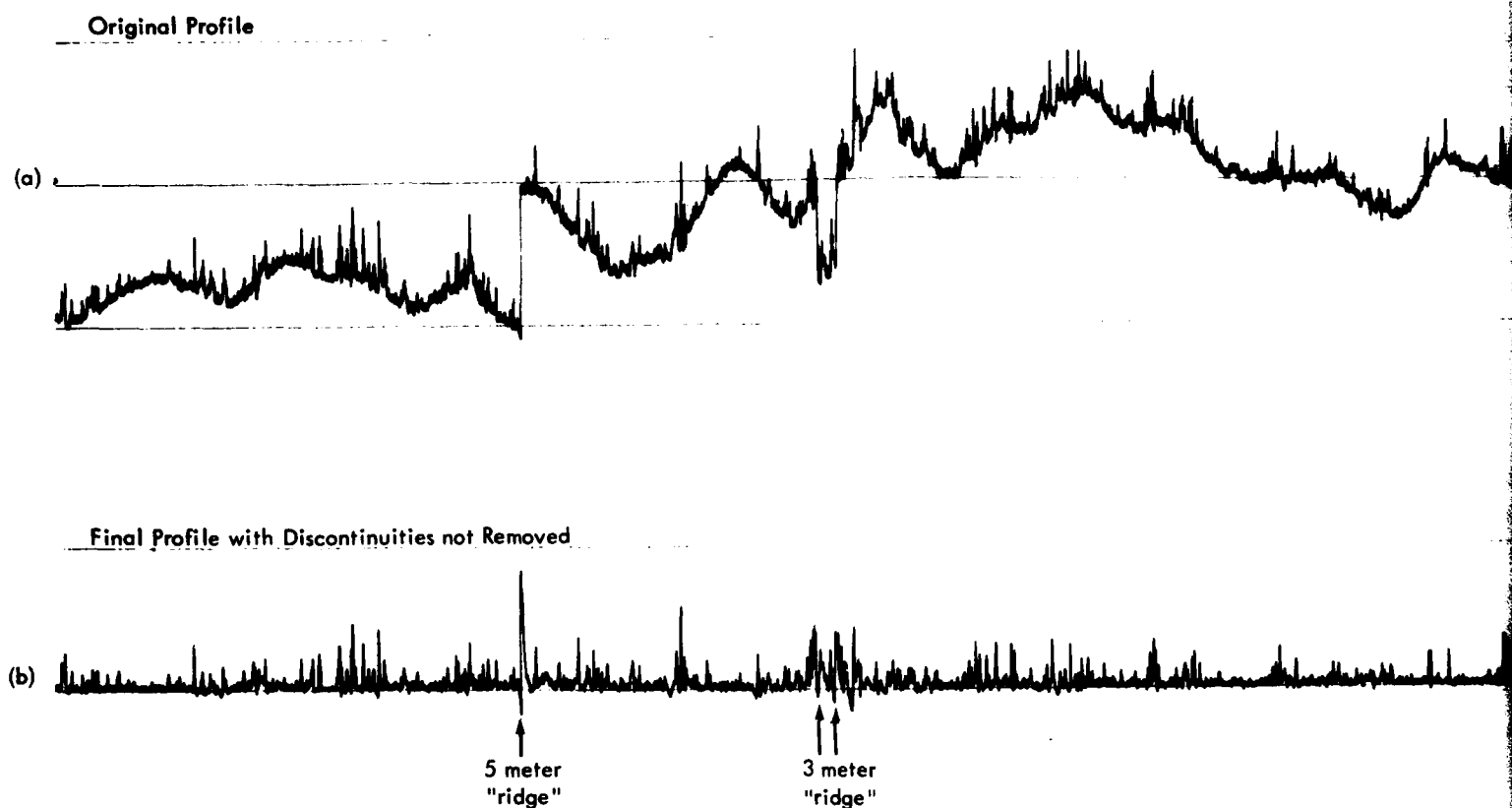
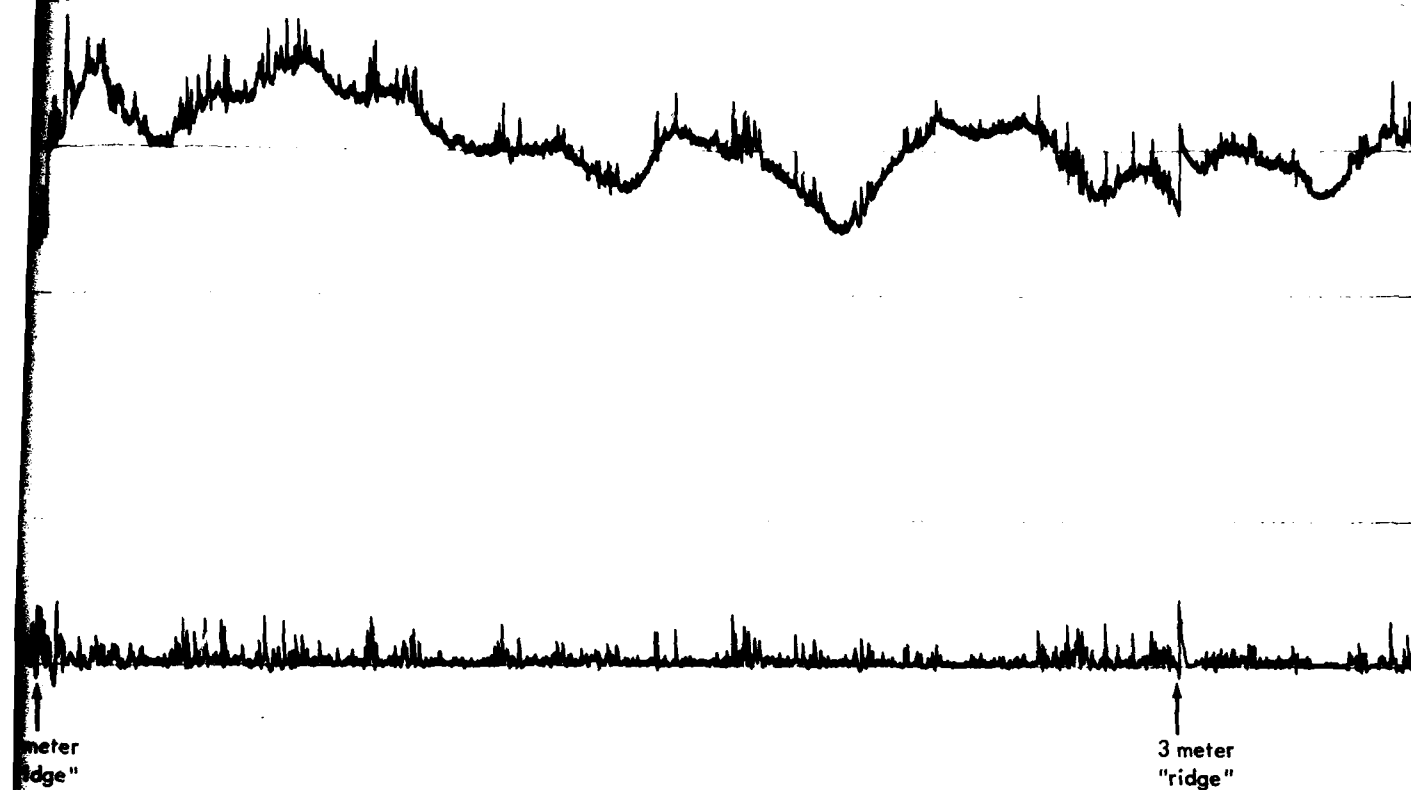


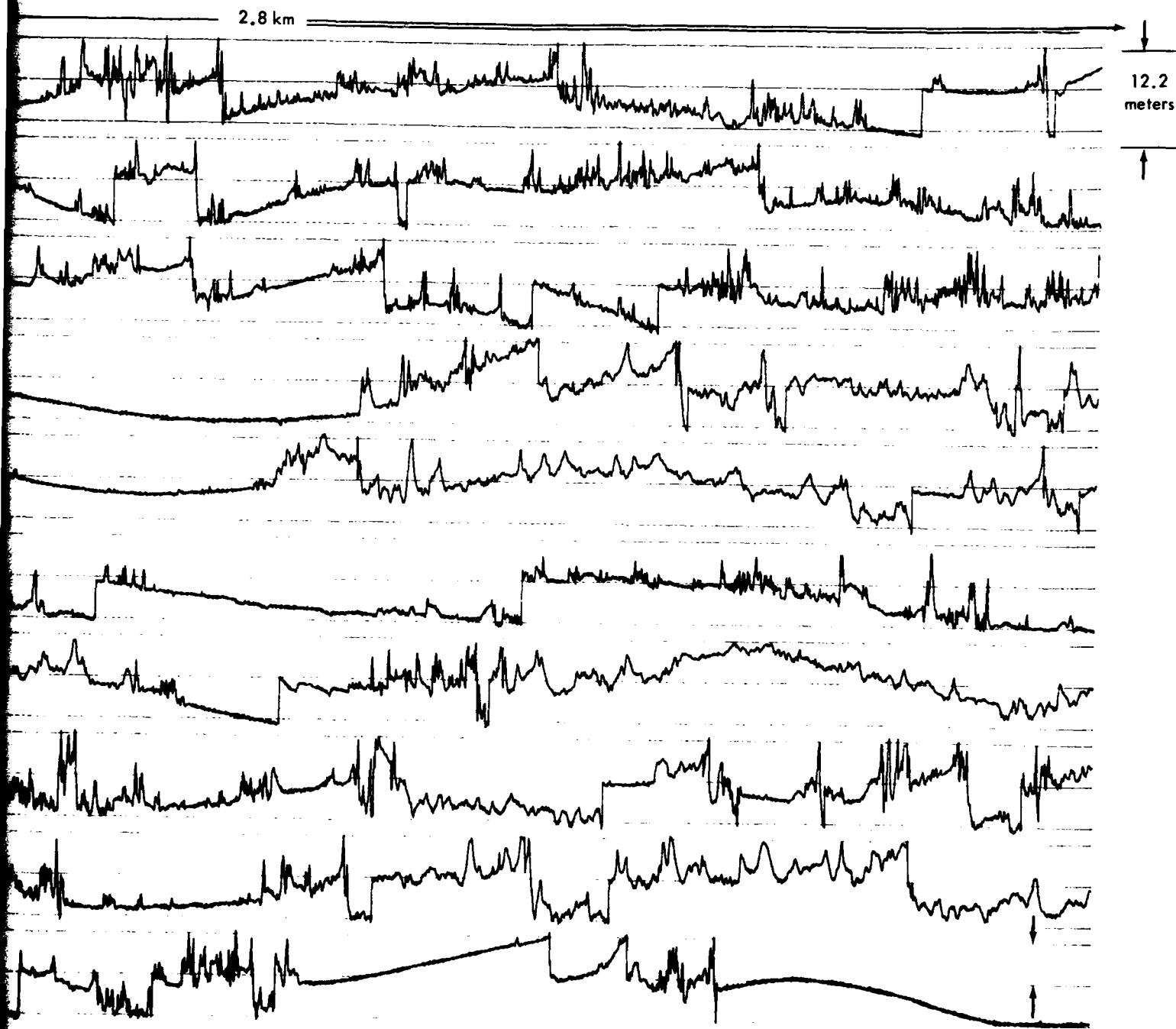
Figure 2. Laser sea-ice profile showing effect of not removing discontinuities. Vertical exaggeration 300:1. Compare with Figure 1(c).



Laser sea-ice profile showing effect of not removing discontinuities.  
Vertical exaggeration 300:1. Compare with Figure 1(c).



Figure 3. Laser sea-ice profile with many discontinuities. Circled area is one of several in which two or three discontinuities occur in rapid succession.



3. Laser sea-ice profile with many discontinuities. Circled area is one of several in which two or three discontinuities occur in rapid succession.

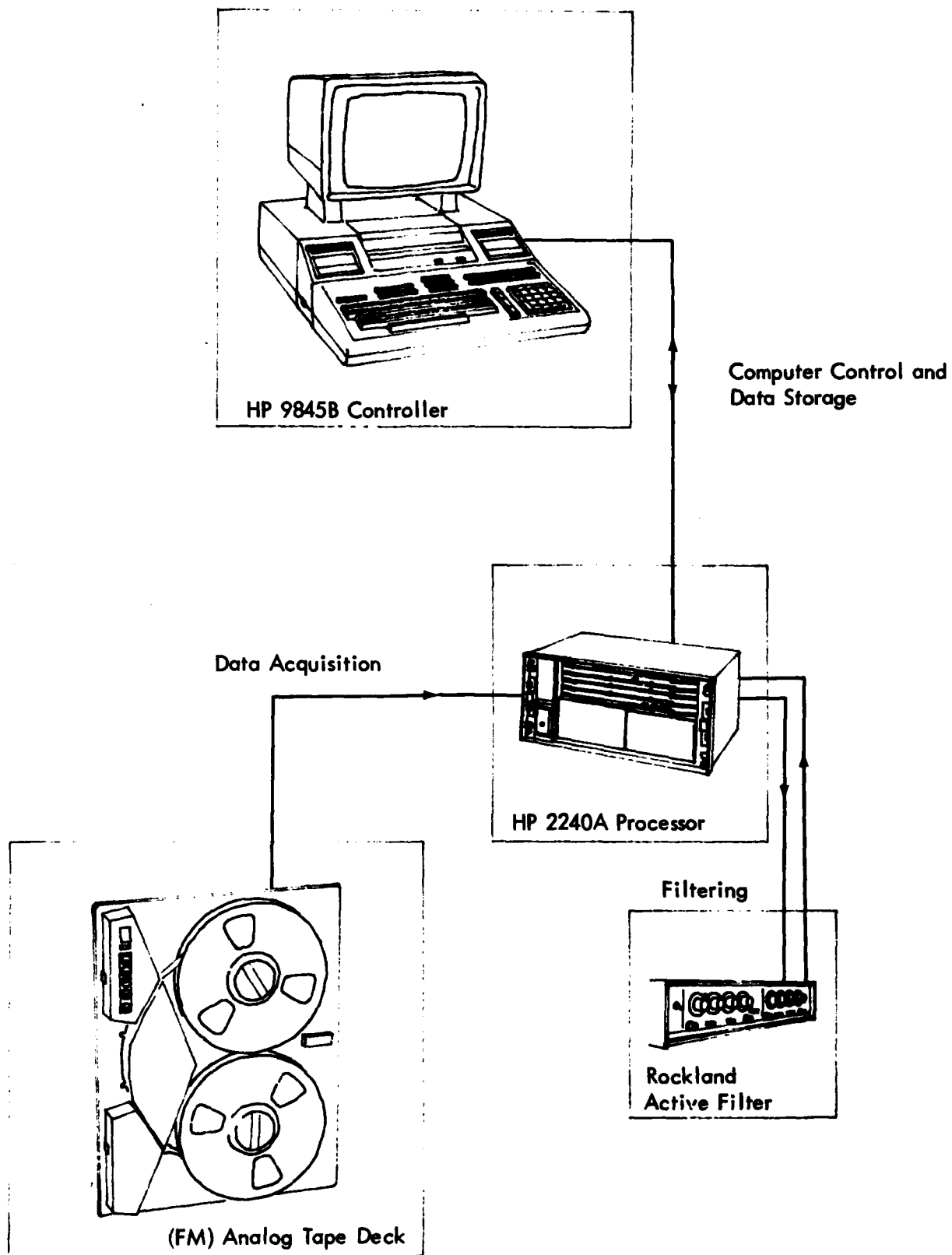


Figure 4. Physical diagram of computer used in laser profile reduction (disc storage not shown)



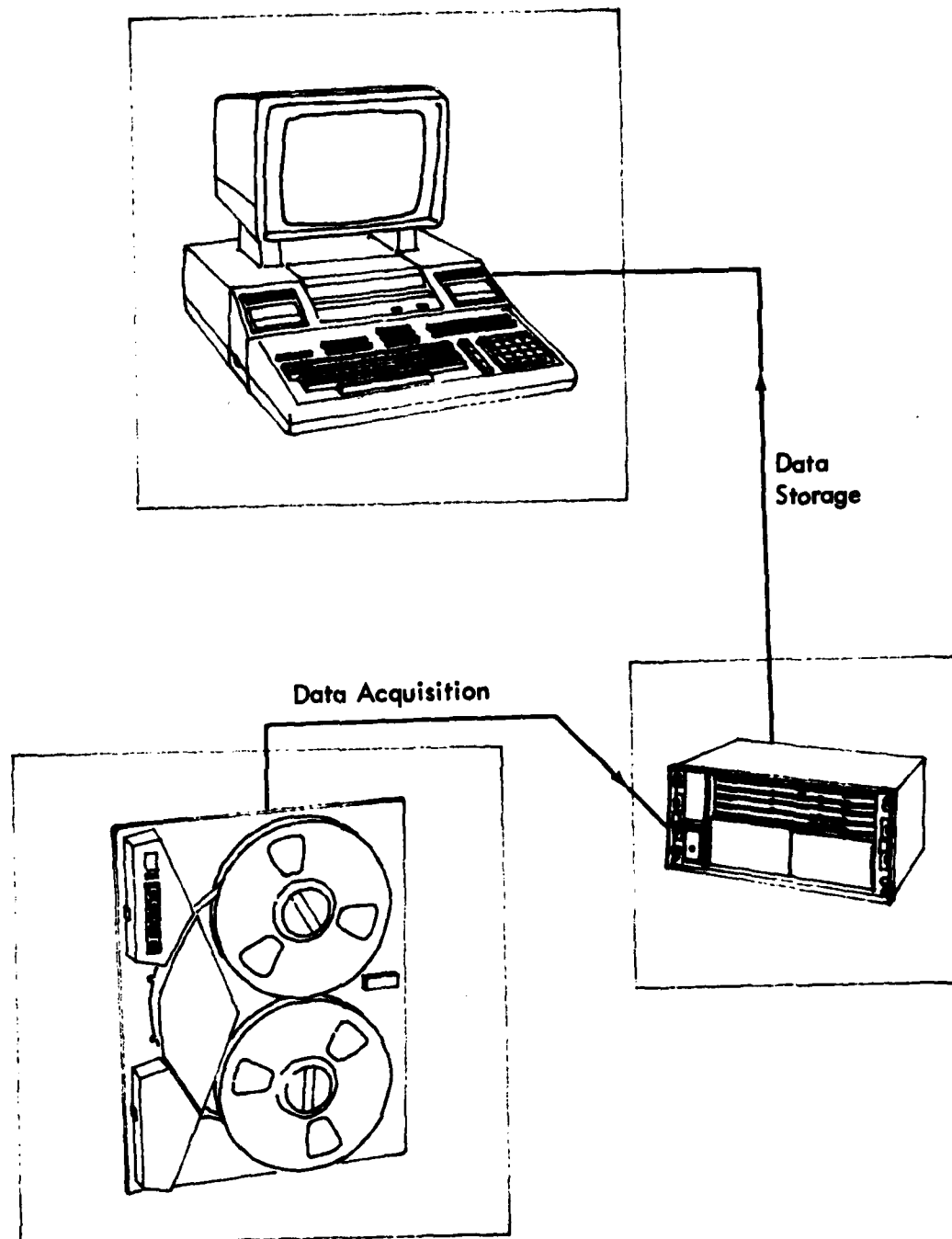


Figure 5. Data acquisition

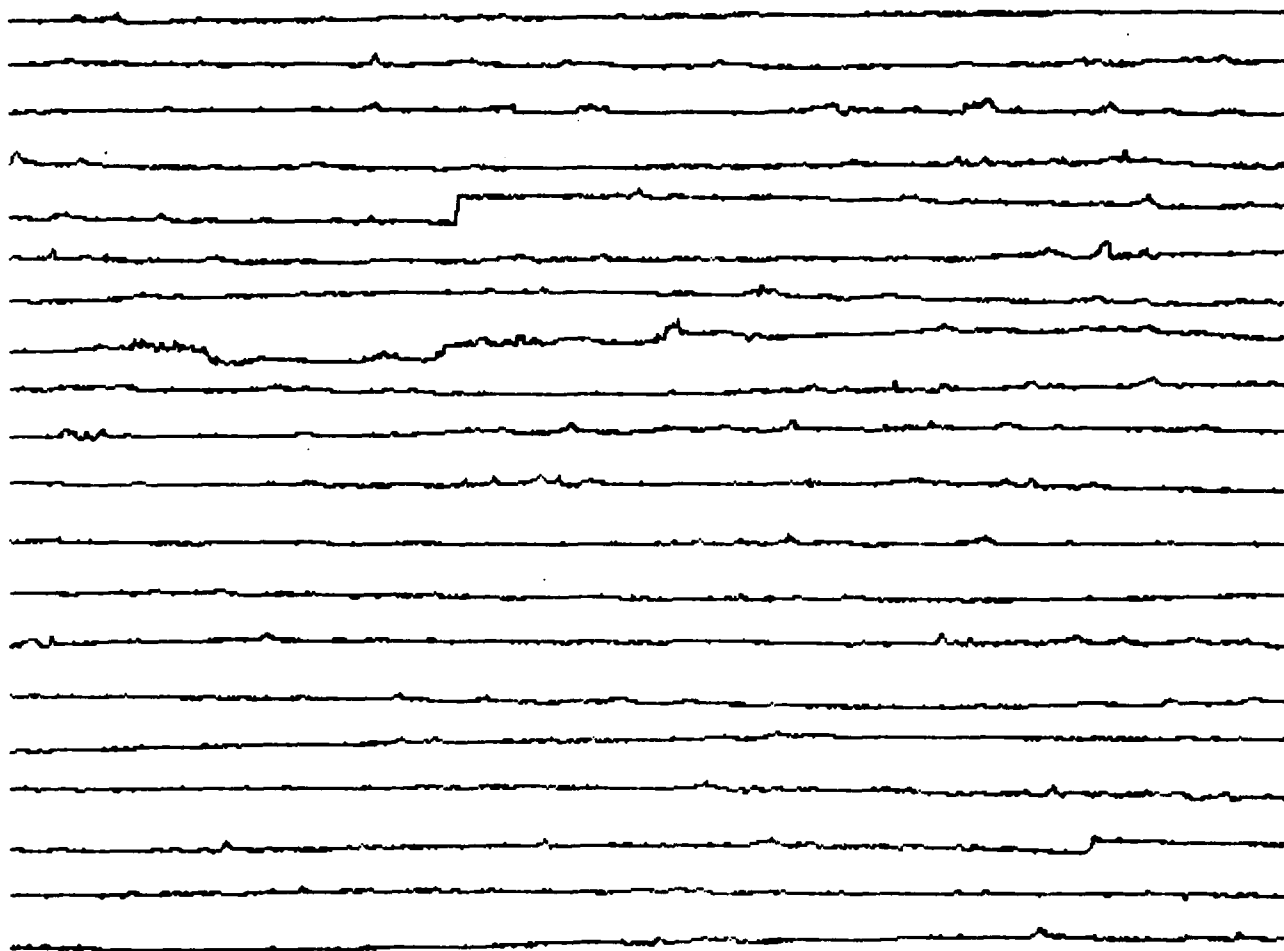


Figure 6. Small-scale CRT plot of 5-minute laser profile record. Shown actual size.

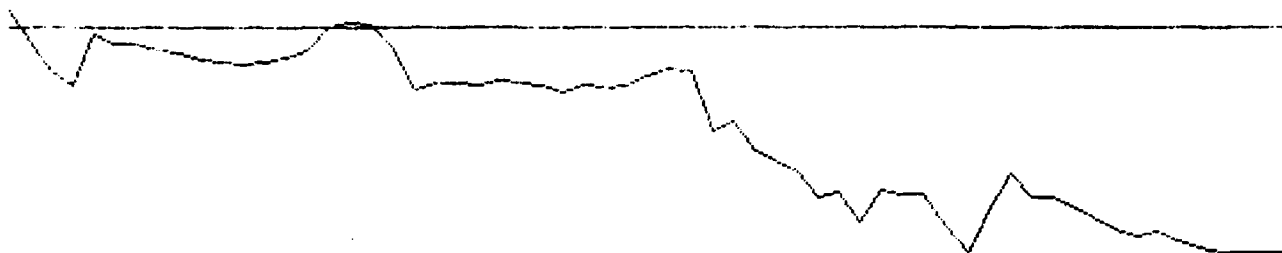
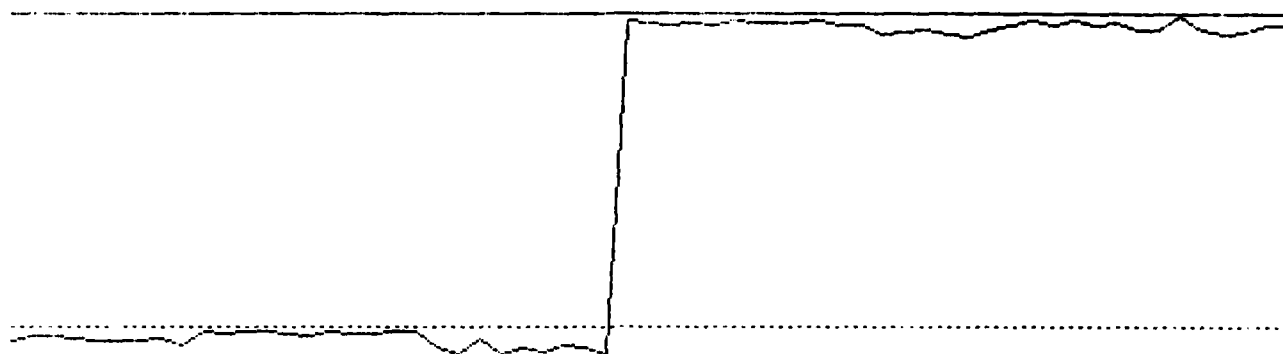
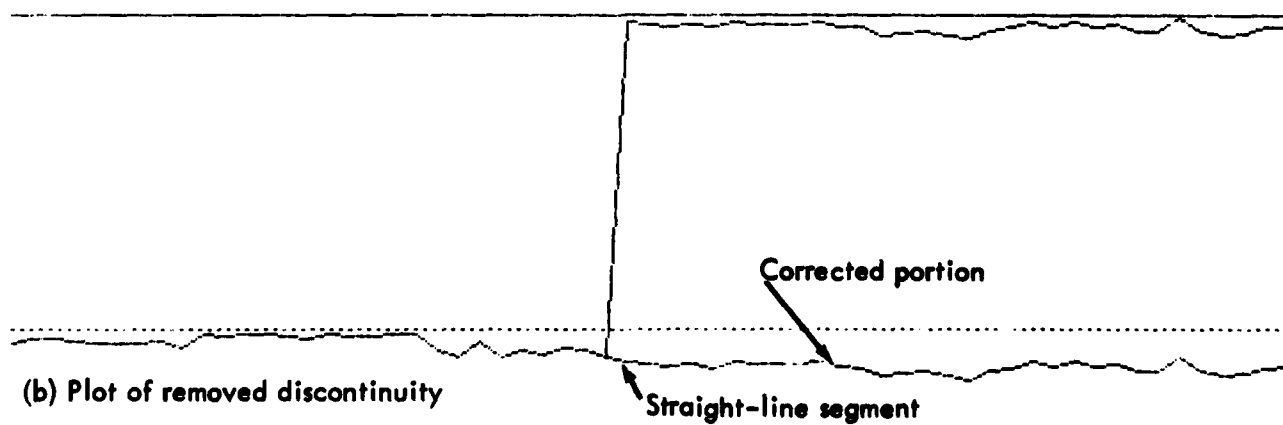


Figure 7. Large-scale plot of 60 points in the vicinity of a profile discontinuity. Shown actual size. Vertical exaggeration 2.25:1.

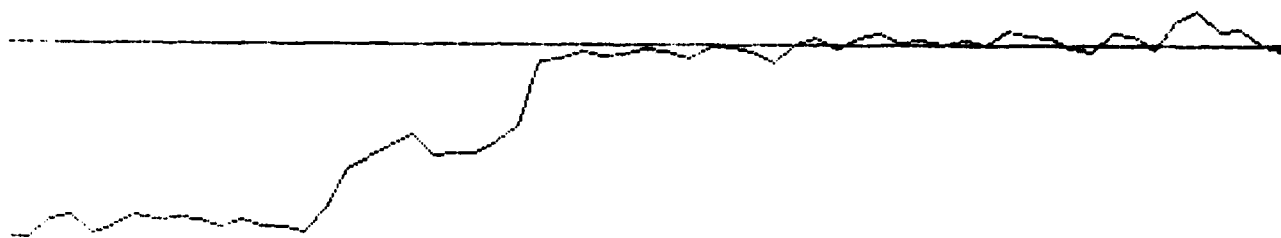


(a) Plot of discontinuity as presented on CRT



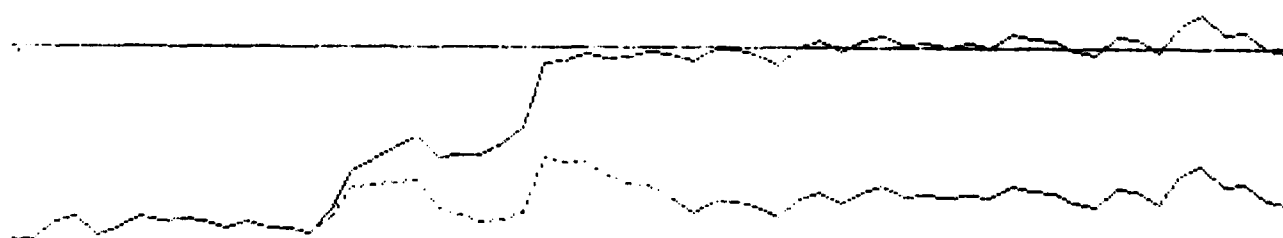
(b) Plot of removed discontinuity

Figure 8. Straight-line editing of a profile discontinuity on the CRT.  
Shown actual size. Vertical exaggeration of profile 2.25:1.



(a) Plot of discontinuity as presented on CRT

STRAIGHT-LINE SLOPE



(b) Plot of removed discontinuity

Figure 9. Slope removal of a profile discontinuity on the CRT shown actual size. Vertical exaggeration of profile 2.25:1.

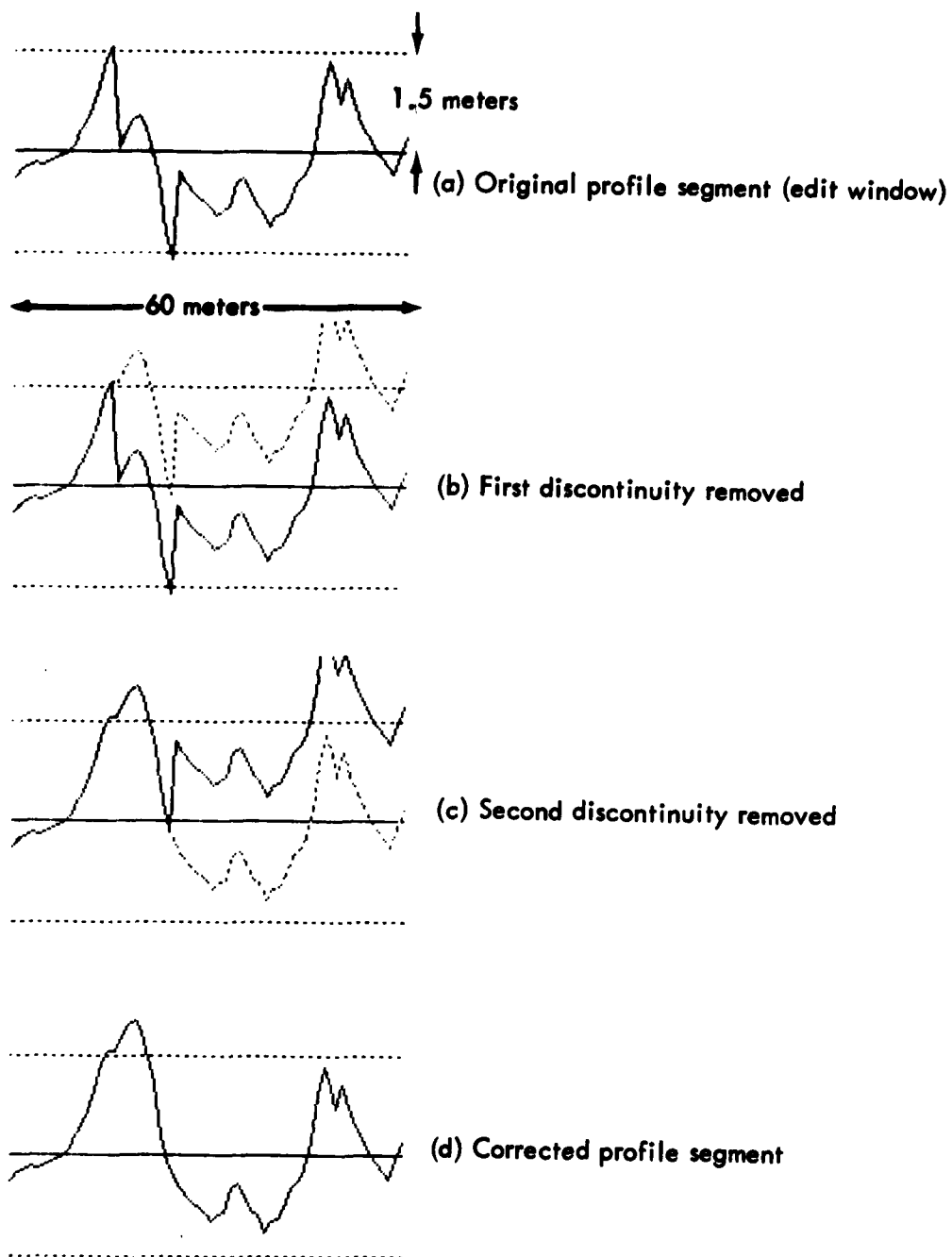


Figure 10. Point-by-point removal of discontinuities from sea-ice profile. Each frame actually fills CRT. Vertical scale exaggeration 10:1.

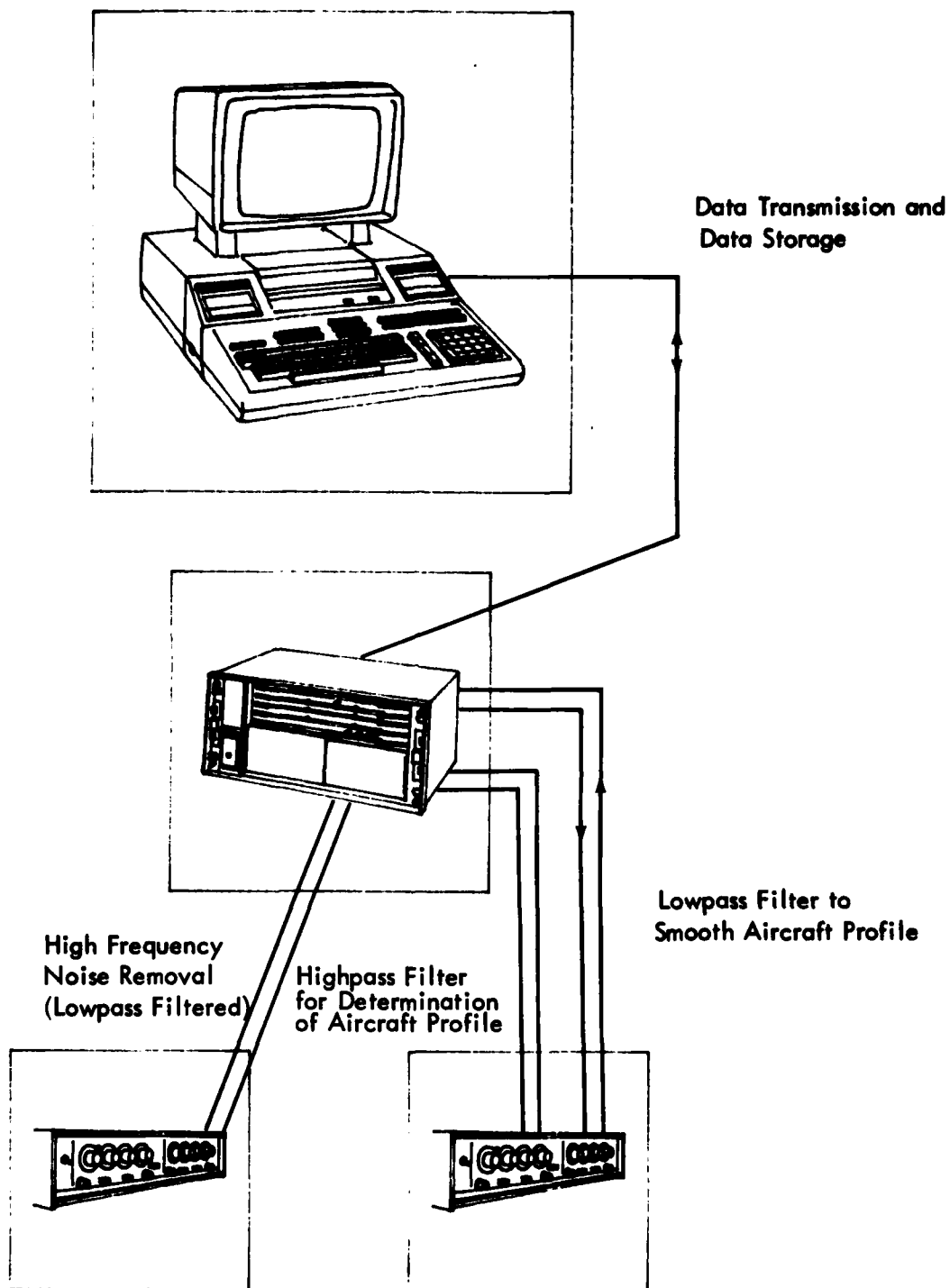


Figure 11. Actively filtering laser profiles

Final profile

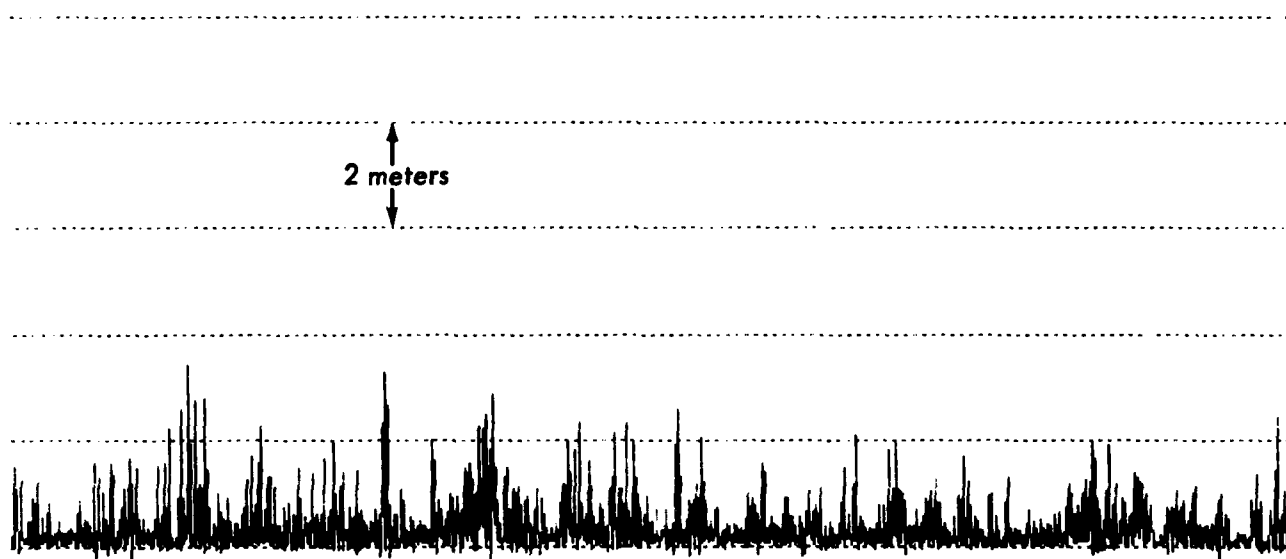
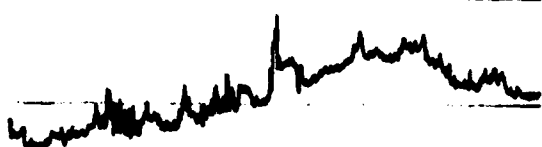


Figure 12. Printout of scaled final sea-ice profile. Vertical scale exaggeration of profile  $\sim 150:1$ .

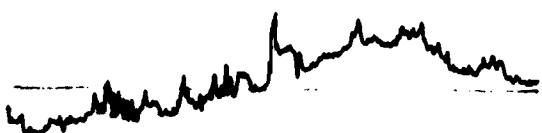




(a) Raw profile data



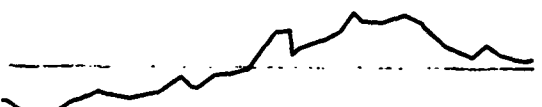
(b) Discontinuities removed



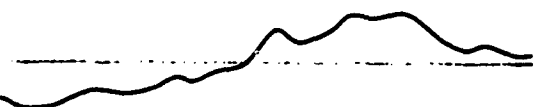
(c) High-frequency noise removed from (b)



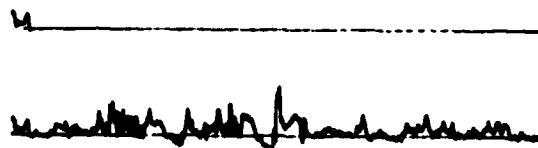
(d) Local low points found by high pass filter of (a)



(e) Low point envelope determined from (a) and (d)



(f) Aircraft motion is lowpass-filtered from low point envelope (e)



(g) Final profile is (c) minus (f)

Figure 13. All steps in aircraft motion removal. Steps (c) through (g) are done automatically by the computer in one program.

**APPENDIX**

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**D**

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1270 PRINT LIN(0)
1280 PRINT " IF SATISFIED WITH EDITING ON THIS WINDOW, enter +"
1290 PRINT " TO RE-EDITING OVER AGAIN FOR THIS WINDOW, enter -"
1300 PRINT " TO CONTINUE THIS WINDOW WITH ANOTHER TYPE OF CORRECTION, enter
/"
1310 Satisfaction=""
1320 INPUT " + or - or /",Satisfaction$
1330 IF (Satisfaction<>"") AND (Satisfaction<>"-") AND (Satisfaction<>"/")
THEN 1320
1340 IF Satisfaction="+" THEN Set_aside
1350 IF Satisfaction="-" THEN Next_try
1360 IF Satisfaction="/" THEN Renew
1361 | <----- WRITE EDIT WINDOW ONTO TAPE CARTRIDGE ----->
1370 Set_aside: | Set aside ( 62 points ).
1380 PRINT 01;X_o
1390 PRINT 01;Zedit(*)
1400 GOTO Next_d
1401 | <----- SUBROUTINE FOR RECORDING BEGINNING OR END OF FILE ----->
1410 SUB Record(INTEGER Xzero,INTEGER Z(*))
1420 GRAPHICS
1430 INTEGER X1,X2,X_1,Z_window(1:61),A
1440 REAL U,V
1450 CSIZE 15/4.54
1460 MOVE 20,90
1470 LABEL "RECORD [ ] END [ ] START"
1480 MOVE 20,85
1490 LABEL "( point , then [ ] )"
1500 POINTER 10,55,0
1501 | <----- PROVIDE CURSOR TO LOCATE BOF or EOF ----->
1510 DIGITIZE X1,Y1
1520 DIGITIZE X2,Y2
1530 IF X1<2 THEN X1=2
1540 IF X1>110 THEN X1=122
1550 X_1=Xzero+INT(X1/2)-1
1560 IF (Y2>09) AND (X2>54) AND (X2<57) THEN Mark=1
1570 IF (Y2>09) AND (X2>76) AND (X2<79) THEN Mark=2
1580 IF Mark=1 THEN Z_window(INT(X1/2))=-32767
1590 IF Mark=2 THEN Z_window(INT(X1/2))=32767
1600 CSIZE 2.5
1610 MOVE X1-10,82
1620 IF Mark=1 THEN LABEL " END | "&VAL$(X_1)
1630 IF Mark=2 THEN LABEL " START | "&VAL$(X_1)
1640 CSIZE 15.4.54
1650 LINE TYPE 3
1660 MOVE X1,81
1670 PLOT X1,81
1680 PLOT X1,19
1690 LINE TYPE 1
1700 PAUSE
1710 PRINTER IS 16
1720 PRINT PAGE
1730 PRINT LIN(5); " If satisfied with the correction, enter +"
1740 PRINT LIN(1); " To re-do the correction, enter -"
1750 Dos=""
1760 INPUT "+ , -",Dos
1770 IF (Dos<>"") AND (Dos<>"-") THEN 1760
1780 IF Dos="-" THEN 1810
1790 Z(INT(X1/2))=Z_window(INT(X1/2)) | CHANGE ONLY THE VALUE OF ONE Z(*)
1800 SUBEXIT
1801 | <----- GRAPH OF THE CORRECTED EDIT WINDOW ----->
1810 GCLEAR
1820 LINE TYPE 3
1830 MOVE 0,80
1840 PLOT 0,80
1850 PLOT 120,80
1860 MOVE 0,20
1870 PLOT 0,20
1880 PLOT 120,20
1890 MOVE 0,50
1900 LINE TYPE 1
1910 PLOT 0,50

```





[illegible]



31

```

580      Loop=1
590      Next_x=Window(2,0)          ! ASSUME Next_x occurs AFTER BOR
600      FOR A=1 TO 61
610          IF Window(1,A)<>32767 THEN 670
620              Print$="ON"
630              Window(1,A)=Window(1,A+1)
640              Current_x=A
650              Window_x=A
660              GOTO Next_point
670      NEXT A
680 Case12:

690      Next_x=Current_x=Window(1,0)
700      Window(1,1)=Window(1,2)
710      GOTO Next_point
720 Case21:

730          Loop=Loop+1
740          IF Loop=No_of_windows THEN Case211
750          Next_x=Window(Loop+1,0)
760          IF Print$="OFF" THEN Skip21
770          Delta_z=Z(Current_x)+Delta_z-Window(Loop,1)
780          IF ABS(Window(Loop,1)+Delta_z)<30000 THEN 870
790          ! ~~~~~
800              REDIM Z(1:X)
810              CALL Record_record(Z(*),X)
820              REDIM Z(1:Z_size)
830          IF Window(Loop,1)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
840          IF Window(Loop,1)+Delta_z>30000 THEN Delta_z=Delta_z-30000
850          X=0
860          ! ~~~~~
870          X=X+1
880          Z(X)=Window(Loop,1)+Delta_z
890 Skip21:          ! DO NOT RE-COMPUTE Delta_z
900          Window_x=2
910          Current_x=Current_x+1
920          GOTO Next_point
930 Case22:

940          Loop=Loop+1
950          IF Loop=No_of_windows THEN Case221
960          Next_x=Window(Loop+1,0)
970          IF Print$="OFF" THEN Skip22
980          Delta_z=Window(Loop-1,Window_x)+Delta_z-Window(Loop,1)
990          IF ABS(Window(Loop,1)+Delta_z)<30000 THEN 1080
1000         ! ~~~~~
1010             REDIM Z(1:X)
1020             CALL Record_record(Z(*),X)
1030             REDIM Z(1:Z_size)
1040         IF Window(Loop,1)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
1050         IF Window(Loop,1)+Delta_z>30000 THEN Delta_z=Delta_z-30000
1060         X=0
1070         ! ~~~~~
1080         X=X+1
1090         Z(X)=Window(Loop,1)+Delta_z
1100 Skip22:          ! DO NOT RE-COMPUTE Delta_z
1110         Window_x=2
1120         Current_x=Current_x+1
1130         GOTO Next_point
1140 Case211:

1150         Delta_z=Z(Current_x)+Delta_z-Window(Loop,1)
1160         GOTO Go_on_21
1170 Case221:

1180         Delta_z=Window(Loop-1,Window_x)+Delta_z-Window(Loop,1)
1190 Go_on_21:

1200         FOR A=1 TO 61
1210             IF Window(Loop,A)=-32767 THEN Skip_21

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1220          REDIM Z(1:X)
1230          CALL Record_record(Z(*),X)
1240          PRINT PAGE
1250          PRINT LIN(5)
1260          PRINT "PROGRAM COMPLETED"
1270          STOP
1280 Skip_21:
1290          IF ABS(Window(Loop,A)+Delta_z)<30000 THEN 1380
1300          ! .....
1310          REDIM Z(1:X)
1320          CALL Record_record(Z(*),X)
1330          REDIM Z(1:Z_size)
1340          IF Window(Loop,A)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
1350          IF Window(Loop,A)+Delta_z>30000 THEN Delta_z=Delta_z-30000
1360          X=0
1370          ! .....
1380          X=X+1
1390          Z(X)=Window(Loop,A)+Delta_z
1400          NEXT A
1410          REDIM Z(1:X)
1420          CALL Record_record(Z(*),X)
1430          PRINT PAGE
1440          PRINT LIN(5)
1450          PRINT "PROGRAM COMPLETED"
1460          STOP
1470 Case3:
1480          IF Window(Loop,Window_x)=32767 THEN Case31
1490          IF Window(Loop,Window_x)=-32767 THEN Case32
1500          GOTO Case_3_continue
1510 Case31:
1520          Print$="ON"
1530          Window(Loop,Window_x)=Window(Loop,Window_x+1)
1540          Delta_z=0
1550          GOTO Case_3_continue
1560 Case32:
1570          Print$="OFF"
1580          Delta_z=0
1590          REDIM Z(1:X)
1600          CALL Record_record(Z(*),X)
1610          X=0
1620          REDIM Z(1:Z_size)
1630 Case_3_continue:
1640          IF Print$="OFF" THEN No_print_3
1650          IF ABS(Window(Loop,Window_x)+Delta_z)<30000 THEN 1740
1660          ! .....
1670          REDIM Z(1:X)
1680          CALL Record_record(Z(*),X)
1690          REDIM Z(1:Z_size)
1700          IF Window(Loop,Window_x)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
1710          IF Window(Loop,Window_x)+Delta_z>30000 THEN Delta_z=Delta_z-30000
1720          X=0
1730          ! .....
1740          X=X+1
1750          Z(X)=Window(Loop,Window_x)+Delta_z
1760 No_print_3:
1770          Current_x=Current_x+1
1780          Window_x=Window_x+1
1790          GOTO Next_point
1800 Case4:
1810          IF Print$="OFF" THEN Skip4
1820          Delta_z=Window(Loop,A)+Delta_z-2*Current_x
1830          IF ABS(2*Current_x+Delta_z)<30000 THEN 1920
1840          ! .....

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1850          REDIM Z(1:X)
1860          CALL Record_record(Z(*),X)
1870          REDIM Z(1:Z_size)
1880      IF Z(Current_x)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
1890      IF Z(Current_x)+Delta_z>30000 THEN Delta_z=Delta_z-30000
1900          X=0
1910      ! ! ! ! !
1920          X=X+1
1930          Z(X)=Z(Current_x)+Delta_z
1940 Skip4:

1950          Window_x=0
1960          Current_x=Current_x+1
1970          GOTO Next_point
1980 Case5:

1990          FOR A=Current_x TO Next_x-1
2000              IF Print#="OFF" THEN 2120
2010              IF ABS(Z(A)+Delta_z)>30000 THEN 2100
2020      ! ! ! ! !
2030              REDIM Z(1:X)
2040              CALL Record_record(Z(*),X)
2050              REDIM Z(1:Z_size)
2060              IF Z(A)+Delta_z<-30000 THEN Delta_z=Delta_z+30000
2070              IF Z(A)+Delta_z>30000 THEN Delta_z=Delta_z-30000
2080              X=0
2090      ! ! ! ! !
2100              X=X+1
2110              Z(X)=Z(A)+Delta_z
2120          NEXT A
2130          Current_x=Next_x
2140          GOTO Next_point
2150      END
2151      ! <<<<< SUBROUTINE FOR STORING THE DATA WITH DISCONTINUITIES REMOVED>>>>>
2160      SUB Record_record(INTEGER Z_in(*),INTEGER N)
2170      PRINTER IS 16
2180      PRINT PAGE
2190      PRINT LIN(8)
2200      PRINT " Name the record which will receive this data."
2210      PRINT LIN(1)
2220      PRINT " LENGTH : ";N;"points."
2230      F$=" "
2240      LINPUT "          XXXXXX:F or XXXXXX:T14",F$
2250      IF (F$(8;1)<>"F") AND (F$(8;1)<>"T") THEN 2240
2260      INTEGER Z(1:N)
2270      MAT Z=Z_in
2280      IF F$(8;1)<>"F" THEN 2320
2290          FCREATE F$,INT(2*N/256)+INT(2*N-65536)+3
2300          FPRINT F$,Z(*)
2310          GOTO 2360
2320      ! DATA file:
2330          CREATE F$,N+4
2340          ASSIGN #1 TO F$
2350          PRINT #1;Z(*)
2360      SUBEND

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1220 IF Interval=1 THEN Delta_z=Zmin(1)-Z_raw(1)
1230 !
1240 IF (Interval>1) AND (Flag=0) THEN Delta_z=Zmin(Interval)-Zmin(Interval-1)
1250 !
1260 IF Flag=1 THEN Delta_z=Z_raw(Observations)-Zmin(Interval-1)
1270 !
1280 Delta_x=Bstop-Bstart+1
1290 FOR B=Bstart TO Bstop
1300 IF Interval=1 THEN Z(B)=Z_raw(1)+Delta_z/Delta_x*(B-Bstart+1)
1310 IF Interval>1 THEN Z(B)=Zmin(Interval-1)+Delta_z/Delta_x*(B-Bstart+1)
1320 NEXT B
1330 IF Flag=0 THEN 1130
1340 SUBEND
1350 SUB Graphics(Q$,P$,INTEGER Z(*),Obs)
1360 INTEGER A,B
1370 PLOTTER IS "GRAPHICS"
1380 GRAPHICS
1390 LINE TYPE 3
1400 FOR B=10 TO 60 STEP 10
1410 MOVE 0,B
1420 PLOT 0,B
1430 PLOT 120,B
1440 NEXT B
1450 LINE TYPE 1
1460 MOVE 0,85
1470 LABEL " Final profile "&P$&"(from "&Q$&") ";Obs;"points."
1480 FOR A=1 TO Obs STEP 10
1490 U=.004*A
1500 V=Z(A)/38.65+10
1510 IF A=1 THEN MOVE U,V
1520 PLOT U,V
1530 NEXT A
1540 DUMP GRAPHICS
1550 PRINTER IS 0
1560 PRINT LIN(5) ! FEED SOME PAPER WITH PLOT
1570 PRINTER IS 16
1580 SUBEND
1590 SUB Active_filter_1(INTEGER Z(*),Observations)
1600 INTEGER Z_store(1:66),Status,A,B,C,D
1610 OUTPUT 7,1;"AC!"
1620 ! PREPARE STEADY VOLTAGE FOR INPUT
1630 FOR A=1 TO 1000
1640 OUTPUT 7,1;"AB,2,1,1,"&VAL$(Z(1))&!"
1650 NEXT A
1660 !
1670 ! INPUT PROFILE
1680 !
1690 FOR B=1 TO Observations
1700 OUTPUT 7,1;"AB,2,1,1,"&VAL$(Z(B))&";AI,1,1,1!"
1710 ENTER 7,1 BFHS 2 NOFORMAT;Status
1720 ENTER 7,1 BFHS 2 NOFORMAT;Z(B)
1730 NEXT B
1740 ! CAPTURE POINTS DUE TO PHASE LAG
1750 FOR C=1 TO 66
1760 OUTPUT 7,1;"AB,2,1,1,"&VAL$(Z(Observations))&";AI,1,1,1!"
1770 ENTER 7,1 BFHS 2 NOFORMAT;Status
1780 ENTER 7,1 BFHS 2 NOFORMAT;Z_store(C)
1790 NEXT C
1800 SUBEND
1810 SUB Active_filter_2(INTEGER Z(*),Observations)
1820 INTEGER Z_store(1:66),Status,A,B,C,D
1830 OUTPUT 7,1;"AC!"
1840 ! PREPARE STEADY VOLTAGE FOR INPUT
1850 FOR A=1 TO 1000
1860 OUTPUT 7,1;"AB,2,2,1,"&VAL$(Z(1))&!"
1870 NEXT A
1880 !
1890 ! INPUT PROFILE
1900 !
1910 FOR B=1 TO Observations

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1920         OUTPUT 7,1;"AB,2,2,1,"&VAL$(Z(B))&";AI,1,2,1!"
1930         ENTER 7,1 BFHS 2 NOFORMAT;Status
1940         ENTER 7,1 BFHS 2 NOFORMAT;Z(B)
1950     NEXT B
1960 ! CAPTURE POINTS DUE TO PHASE LAG
1970     FOR C=1 TO 66
1980         OUTPUT 7,1;"AB,2,2,1,"&VAL$(Z(Observations))&";AI,1,2,1!"
1990         ENTER 7,1 BFHS 2 NOFORMAT;Status
2000         ENTER 7,1 BFHS 2 NOFORMAT;Z_store(C)
2010     NEXT C
2020     SUBEND
2030 SUB Active_filter_3(INTEGER Z(*),Observations)
2040     INTEGER Z_store(1:66),Status,A,B,C,D
2050     OUTPUT 7,1;"AC!"
2060 ! PREPARE STEADY VOLTAGE FOR INPUT
2070     FOR A=1 TO 1000
2080         OUTPUT 7,1;"AB,2,3,1,"&VAL$(Z(1))&";!"
2090     NEXT A
2100 !
2110 ! INPUT PROFILE
2120 !
2130     FOR B=1 TO Observations
2140         OUTPUT 7,1;"AB,2,3,1,"&VAL$(Z(B))&";AI,1,3,1!"
2150         ENTER 7,1 BFHS 2 NOFORMAT;Status
2160         ENTER 7,1 BFHS 2 NOFORMAT;Z(B)
2170     NEXT B
2180 ! CAPTURE POINTS DUE TO PHASE LAG
2190     FOR C=1 TO 66
2200         OUTPUT 7,1;"AB,2,3,1,"&VAL$(Z(Observations))&";AI,1,3,1!"
2210         ENTER 7,1 BFHS 2 NOFORMAT;Status
2220         ENTER 7,1 BFHS 2 NOFORMAT;Z_store(C)
2230     NEXT C
2240     FOR D=1 TO Observations
2250         IF D<=Observations-66 THEN Z(D)=Z(D+66)
2260         IF D>Observations-66 THEN Z(D)=Z_store(D-(Observations-66))
2270     NEXT D
2280     SUBEND
2290 SUB Warning
2300     PRINT PAGE
2310     PRINT " Aircraft motion being removed."
2320     PRINT LIN(1)
2330 PRINT " "
2340 PRINT " "
2350 PRINT " "
2360 PRINT " "
2370 PRINT " "
2380 PRINT " "
2390 PRINT " "
2400 PRINT " "
2410 PRINT " "
2420 PRINT " "
2430 PRINT " "
2440 PRINT " "
2450 PRINT " "
2460 PRINT " "
2470 PRINT " "
2480 PRINT " "
2490 PRINT " "
2500 SUBEND

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